Milestone Review Flysheet

PDR, CDR, FRR

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Vehicle Properties		
Diameter	6.16	
(inches)		
Length	121	
(inches)		
Gross Liftoff Weight	39.3	
(pounds)		
Lounch lug/button size	Public Missile Linear Launch Rail	
Launch lug/button size	Lugs	
	Currently under review	
Motor Retention Method		

Stability Analysis	
Center of Pressure	100.7
(inches from nose)	
Center of Gravity	68.19
(inches from nose)	
Thrust-to-Weight Ratio	6.79
Rail Size/Length	1 inch by 8 feet
Rail Exit Velocity	Under research
(feet/second)	

Recovery System Properties	
Dr	ogue Parachute
Manufacturer/Model	SkyAngle Classic 44"
Size	44"
Altitude at Deployment	5305
(feet)	
Velocity at Deployment	1.03
(feet/second)	

Recovery System Properties		
Electronics/Ejection		
Altimeter(s) Make/Model	PerfectFlight Mini Altimeter	
Redundancy Plan	2 altimeters on board	
Pad Stay Time (Launch	3 Minutes	

Motor Properties	
Motor Manufacturer(s)	Contrail Rockets
Motor Designation(s)	L-1222-SM
Max, Average Thrust (Newtons)	Maximum: 2540 Average: 1191.7
Total Impulse (Newton-seconds)	3694
Mass before/after burn (pounds)	Before: 39.3 After: 30.7

Ascent Analysis	
Max Velocity	622
(feet/second)	
Max Mach Number	0.55
Max Acceleration	484
(feet/second^2)	
Peak Altitude	5305
(feet)	

Recovery System Properties	
M	ain Parachute
Manufacturer/Model	Public Missile 120"
Size	120"
Altitude at Deployment	800
(feet)	
Velocity at Deployment	61.84
(feet/second)	
Velocity at Landing	between 17 and 22
(feet/second)	

Recovery System Properties	
Elec	ctronics/Ejection
Rocket Locators (Make, Model)	Walston Retreival System
Transmitting Frequencies	***Required by CDR***
Black Power Mass Main Parachute (grams)	Under research
Black Power Mass	Under research

Configuration)	Main Parachute (grams)
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Institution Name	Harding University	Milestone	PDR

Payload/Science		
Succinct Overview of Payload/Science Experiment	We will take data for the temperature, humidity, pressure, solar irradiance, and ultraviolet radiation during descent and after landing. We will also take three pictures during descent and three pictures after the payload has landed. We intend to separate the payload from the airframe at 2,500 feet.	
Identify Major Components	 Payload pod 2. Camera 3. Humidity Sensor 4. Temperature Sensor 5. Pressure Sensor Ultraviolet Radiation Sensor 7. Solar Irradiance Sensor 8. Payload Computer 9. Payload Parachute 10. GPS Tracking System 	
Mass of Payload/Science	Currently Under Research	

Test Plan Schedule/Status					
Ejection Charge Test(s)	Testing to be done in February				
Sub-scale Test Flights	Test flight set for January				
Full-scale Test Flights	Test fligt set for March				

	Additional Comments							
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